

**Research Article** 

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## Comparative analysis and optimization of thermodynamic behavior of combined gas-steam power plant using grey-taguchi and artificial neural network

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## ABSTRACT

In the published studies, to the best of the authors' understanding, the grey Taguchi-based statistical technique has not been applied for the optimization of combined gas-steam power plants. In view of this, seven essential input parameters namely compressor inlet air temperature, pressure ratio, fuel temperature, volumetric flow rate of fuel, gas turbine maximum temperature, compressor efficiency, and turbine efficiency are chosen with the aim of determining the optimal combination of design variables that maximize the net power generation, thermal efficiency, exergetic effciency, and minimize the specific fuel consumption. Also, the impact weight of each parameter on output indicators has been evaluated. While the Taguchi approach helps to create an orthogonal array of L27  $(3^7)$ , the ANOVA method determines the contribution of each input argument on the objective function. Unlike the Taguchi and ANOVA optimization methodology, the grey relational analysis is performed to transform the multi-objective function into a single objective by way of estimating its grey relational grade. The most favorable combination of input parameters is determined as A1B1C1D1E3F3G3 and under this state, the optimum values of power generation, thermal efficiency, exergetic efficiency, and specific fuel consumption are found to be 259911 kW, 64.9 %, 66.27 %, and 0.1839 kg/ kWh respectively. Moreover, the contribution ratio on the output characteristic of the combined cycle is found to be maximum for turbine efficiency (42.41 %) and minimum for fuel temperature (0.59 %). The effectiveness of the grey-Taguchi method is acknowledged and validated using an artificial neural network technique in MATLAB.

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## INTRODUCTION

Presently, in developing countries, economic growth is most desirable, and energy is an essential requirement for economic growth. The increasing energy demand for economic growth is the major issue and challenge that draws appreciable attention towards energy conversion systems. Combined gas-steam-based power plants provide an efficient and effective technology for the conversion of energy resources with high fuel to power efficiency and low environmental emissions. Although, renewable energy sources cause no or low environmental emissions, they cannot generate so much electricity to cater for the complete energy demand. The latest data released by the International Energy Agency (IEA) [1] reveal that the contribution of all the renewables taken together to the total worldwide gross production of electricity is 25.6 % while the contribution of coal-based thermal power plants is 38.2 % and that of natural gas-based thermal power plants is 23.1 %. Therefore, the renewable energy-based power generation systems cannot completely replace the conventional thermal power plants although they are gaining increasing popularity. The natural gas-based thermal power plants basically use gas turbines to generate power. Combined cycle (CC) is a technology widely used these days for efficiency augmentation of the basic gas turbine plants. The most common CC is referred to as the Brayton-Rankine power generating cycle that utilizes heat energy of natural gas to produce power from two individual cycles. Many researchers conducted the performance test of combined cycle-based power plants theoretically and experimentally aiming to enhance the first and second law efficiency. The exergy method is a means to evaluate the energy quality, pinpoint the true location and magnitude of actual losses, and determine the maximal performance of a system using exergetic efficiency and exergy destruction rate [2,3]. Ersayin and Ozgener [4] examined the thermodynamic performance behavior of a CCPP in terms of energy and exergy efficiencies by utilizing the operational statistics collected from the power station control unit. Singh and Kaushik [5] performed the exergybased assessment of a 135 MW steam-based power plant situated at Delhi and assessed the effect of different operating variables on the performance parameters of the power plant through MATLAB simulation. The boiler of the plant was identified as the component of maximum exergy destruction. Singh [6] assessed thermodynamic irreversibility in various sections of a high-pressure boiler to find out the exact location of the maximum exergy destruction in it. In the analysis conducted by Singh and Kaushik [5], it was worked out that the low-temperature heat discharging out from the stack could be utilized to generate additional power. Singh [7,8] also investigated the possibility of utilizing low-grade waste exhaust heat from the cogeneration power plant of a sugar factory for running a cold storage and for generating power through the Kalina cycle.

Madan and Singh [9] utilized the flue gas heat discharging out from the stack to investigate the performance of a low-temperature operated organic Rankine cycle through energy and exergy-based analysis. Singh [10] integrated a waste heat-operated absorption refrigeration unit with the compressor of a 330 MW capacity natural gas-fired combined cycle power plant situated at Delhi to cool the incoming air into the two gas turbines. The outcomes indicated that the power generation of the plant increased by 9440 kW raising the energy efficiency by 1.193 % and exergetic efficiency by 1.133 %. In combined cycle power plants, the thermodynamic performance and the power generation are greatly affected by the gas turbine variables particularly, compressor pressure ratio, compressor inlet air temperature, isentropic efficiencies of compressor and turbine, fuel temperature, air/fuel ratio, exhaust temperature of gas turbine, pinch point temperature difference, inlet pressure and temperature of steam turbine, steam condition at condenser inlet and cooling water flow conditions, stack temperature. Although thermodynamic analysis is of great importance, the optimization of these power plants is mandatory for the continuous growth and development of the power generation sector. A broad spectrum of research has been carried out in literature to analyze and optimize the operation of power generation plants. Pan et al. [11] proposed a waste heat exchanger that includes a supercritical CO<sub>2</sub> cycle, an organic Rankine cycle, and a vapor absorption refrigeration cycle and optimized the integrated system in respect to its thermodynamic, economic, and environmental parameters. Many researchers performed multi-objective optimization of gas-steam cycle applying non-dominated sorting genetic algorithm (NSGA-II) optimization technique for augmenting thermodynamic performance and minimizing exergoeconomic cost, environmental impact, and heat transfer component cost [2,12-18]. Nadir et al. [19] optimized the thermodynamic configuration of the heat recovery steam generator for maximizing the net specific work output of the steam cycle and the net present worth as an objective function using the Particle Swarm Optimization technique. Even though many research articles have been published on the aforementioned optimization techniques, these methods are overly complex, non-linear, and time-intensive. Hence, simple, effective, linear, and less time-consuming methods are to be explored to maximize the performance curve of the studied system.

Grey relation analysis (GRA) is the most prominent and efficient statistical technique that evaluates multiple objective functions with a minimum number of trials [20,21]. Gul et al. [22] conducted an optimization study for a DI-CI diesel engine with different combinations of fuel consumption, speed, and load employing the grey-Taguchi method. Additionally, the ANOVA technique is applied to scrutinize the high impact parameter. The author in his another work [23] also conducted optimization studies on industrial gas turbines with different scales of input parameters using the aforesaid technique. Bademlioglu et al. [24] selected nine parameters to perform optimization and evaluated the weight of each parameter influencing the performance behavior of the organic Rankine cycle using the grey-Taguchi statistical technique. Analogously, the application of the grey-Taguchi optimization methodology is widely applied in materials and industrial sectors [25–29]. The experimental findings and the simulated results obtained through the artificial neural network (ANN) technique with MATLAB software to conduct theoretical studies, e.g., [22] have been mostly found to match well with each other which verifies and validates the use of this technique. It is a technique that develops a highly complex non-linear correlation between input variables and output parameters [30,31].

In literature, the grey relational-Taguchi approach has been frequently applied to numerous industrial processes such as welding, turning, machining, cutting, etc. However, no research study is found regarding its application for optimizing a combined gas-steam power plant. Therefore, in the present work, this technique is chosen to conduct multi-objective optimization of a gas-steam-based power plant to maximize the net power generation, thermal efficiency and exergetic efficiency and minimize the specific fuel consumption simultaneously. In view of this, seven essential input parameters namely the compressor inlet air temperature, pressure ratio, fuel temperature, volumetric flow rate of fuel, gas turbine maximum temperature, compressor efficiency, and turbine efficiency for three different levels are chosen using the grey relational statistical technique. The impact weight of each parameter on performance indicators is then calculated using the analysis of variance (ANOVA) method. Furthermore, the best and worst combinations of input parameters are defined as a result of grey relational analysis, and the results are validated using the artificial neural network approach.

## PLANT DESCRIPTION AND ASSUMPTIONS

In this work, the operating data of a multi-shaft gassteam-based power generation system shown in Fig. 1. are collected from the National Capital Power Station, NTPC, Dadri that generates 415MW of power. The performance specifications of the active CCPP are presented in Table 1. The composition of the fuel used is as follows: 94% CH<sub>4</sub>, 4%  $C_2H_6$ , 1.2%  $C_3H_8$ , 0.8%  $C_4H_{10}$  by volume. The system comprises of an air compressor, a fuel-heated chamber, a high-temperature gas turbine, a dual pressure exhaust heat recovery generator (EHRG), a power generating steam

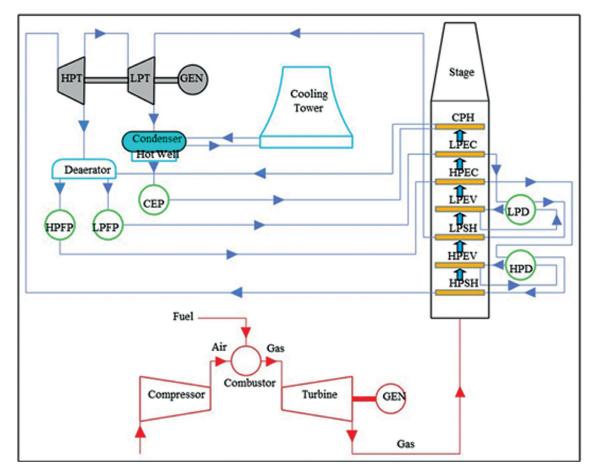


Figure 1. Simplified diagram of Gas-steam power generating plant.

 Table 1. Performance specification of active CCPP

| Input Data                         | Value  | Units |
|------------------------------------|--------|-------|
| Capacity                           | 415    | MW    |
| Gas Turbine Power Output           | 131*2  | MW    |
| Steam Turbine Power Output         | 154    | MW    |
| Combined Cycle Efficiency          | 48.33  | %     |
| Maximum Temperature of Gas Turbine | 1333.2 | Κ     |
| Compressor Pressure Ratio          | 10.2   | -     |
| Condensate Extraction Pump Power   | 270    | kJ/s  |
| Power Utilized by HP BFW Pump      | 1510   | kJ/s  |
| Power Utilized by LP BFW Pump      | 420    | kJ/s  |

turbine (HP & LP), a water-cooled condenser, feedwater pumps and a deaerator. The high-pressure air and fuel combine in the chamber to produce high-temperature products that expand in the turbine to generate shaft work. Both gas turbine and air compressor are linked with common shaft. The high-temperature exhaust gases of the turbine are used by the recovery generator to deliver steam which is further used to drive steam turbine for power generation in combined cycle. The low-temperature combustion products are then disposed into the environment through the stack. The low-temperature wet steam departing from LP turbine is condensed in a water-cooled condenser and the condensate is pumped back to the boiler by the feedwater pump via a deaerator. The temperature of the low-grade heat discharging out from the stack is 100°C.

Following assumptions are used in the present work.

- Steady-state condition is assumed for energy and exergy flow.
- Potential and kinetic energies of the fluid at entry and exit of a plant component are neglected.
- All gases are assumed as ideal.
- The combustion of fuel takes place at constant pressure with no heat transfer to or from the surrounding.
- The working fluids for the topping and the bottoming cycles are the combustion gases and water/steam respectively.
- The expansion and the compression of fluid in the steam turbine and pumps are reversible adiabatic processes.
- Specific heat is a function of temperature only.
- Specific heat ratios for both air and combustion gases are taken as 1.4.

## SYSTEM MODELLING

## Mass, Energy, and Exergy Balance

Applying steady state energy equation to a control volume under given conditions, the mass, energy, and exergy balances can be defined as [5]

$$\sum \widehat{m}_{in} = \sum \widehat{m}_{ex} \tag{1}$$

$$\hat{E}_{Q} - \hat{E}_{W} = \sum \hat{m}_{ex} h_{ex} - \sum \hat{m}_{in} h_{in}$$
(2)

$$\sum \hat{m}_{in} \Psi_{in} + \hat{X}_{Q} = \sum \hat{m}_{ex} \Psi_{ex} + \hat{X}_{W} + \hat{X}_{D}$$
(3)

The rate of flow of exergy  $\hat{X}_Q$  due to heat transfer  $\hat{Q}_i$ at temperature T and that due to work transfer  $\hat{W}$  can be defined as

$$\widehat{X}_{Q} = (1 - \frac{T_{0}}{T})\widehat{Q}_{i} \tag{4}$$

$$\widehat{X}_{w} = \widehat{W} \tag{5}$$

The total specific exergy  $\Psi_{tot}$  comprises physical, chemical, kinetic and potential exergies

$$\Psi_{tot} = (h - h_0) - T_0(s - s_0) + \Psi_{ch} + \frac{c^2}{2} + g_E Z_0 \qquad (6)$$

Neglecting kinetic and potential exergies, it can be written as:

$$\Psi_{tot} = \Psi_{phy} + \Psi_{ch} \tag{7}$$

#### Physical Exergy

The maximum reversible work is gained in a physical process when a substance flows from its initial condition to the atmospheric condition involving thermal interaction only [32].

For water and steam,

$$\Psi_{ph} = (h - h_0) - T_0(s - s_0) \tag{8}$$

For the mixture of ideal gases,

$$\Psi_{ph} = (T - T_0) \sum_{i=1}^m x_i \hat{c}_P^{\Psi} + RT_0 \ln P / P_0$$
(9)

where R stands as the characteristic gas constant,  $T_0$  is the ambient temperature, x is the mole fraction,  $\hat{c}_p^{\Psi}$  is the specific molar heat capacity of exergy and  $P_0$  is the ambient pressure.

For fuel,

The physical exergy of fuel at the environmental state  $(P_0, T_0)$  is zero. However, if the fuel flows into the combustion zone at some other condition, its physical exergy needs to be taken into ccount.

## Chemical exergy

The maximum reversible work is gained when a substance flows from atmospheric condition to the dead state involving reaction and transfer of heat with the environment alone [32]. For Air,

Air mainly comprises of  $N_2$ ,  $O_2$ ,  $CO_2$ ,  $H_2O$  and Ar which are in their elemental state. There is no shift in its chemical composition. Hence, there is no change in the chemical exergy [32] of air.

For a mixture of gases,

$$\Psi_{ch,m} = \sum_{i=1}^{m} x_i \Psi_{ch,i} + R_u T_0 \sum_{i=1}^{m} x_i \ln x_i$$
(10)

#### **Combustion Process**

The volumetric composition of natural gas considered is as follows:

| Fuel component | $\operatorname{CH}_4$ | $C_2H_6$ | $C_{3}H_{8}$ | $C_4H_{10}$ | N <sub>2</sub> |
|----------------|-----------------------|----------|--------------|-------------|----------------|
| % by volume    | 94                    | 4        | 1.2          | 0.78        | 0.02           |

The actual composition of air by volume is calculated at an ambient state ( $P_0$ =100900 Pa,  $T_0$ =307.15 K, RH= 36 %) on the day of data collection using the procedure described in [33]

| Air component | <b>O</b> <sub>2</sub> | N <sub>2</sub> | $H_2O(g)$ | CO <sub>2</sub> | Ar   |
|---------------|-----------------------|----------------|-----------|-----------------|------|
| % by volume   | 20.55                 | 76.61          | 1.89      | 0.03081         | 0.92 |

The Air to fuel ratio by mass may be expressed as

$$(A/F)_{mass} = \frac{n_a \times M_a}{n_f \times M_f} \tag{11}$$

where n stands for number of moles, M is molecular mass, subscript a and f stands for air and fuel.

The rate of heat generation by fuel in kJ/kmol of fuel is given as [34]

$$\dot{E}_{cf} = -\left[\sum_{P} \dot{N}_{P} (\bar{h}_{f,P}^{0} + \bar{h}_{T} - \bar{h}_{298}^{0}) - \sum_{R} \dot{N}_{R} (\bar{h}_{f,R}^{0} + \bar{h}_{T} - \bar{h}_{298}^{0})\right] (12)$$

Following equations are taken from [10] to calculate the specific molar heat capacity  $\hat{c}_{p,a}$  (kJ/kmol-K) for the fuel and other gases, where T is the temperature in K.

$$\hat{c}_{p,CH_4} = 19.25 + 5.213 \times 10^{-2} T + 1.197 \times 10^{-5} T^2 - 1.132 \times 10^{-8} T^3$$
 (13)

$$\hat{c}_{p,C_2H_6} = 5.409 + 17.81 \times 10^{-2} T + 6.938 \times 10^{-5} T^2 + 8.713 \times 10^{-8} T^3 (14)$$

$$\hat{c}_{p,C_3H_8} = 4.224 + 30.63 \times 10^{-2} T + 15.86 \times 10^{-5} T^2 + 3.215 \times 10^{-8} T^3$$
 (15)

$$\hat{c}_{p,C_4H_{10}} = 9.487 + 33.13 \times 10^{-2} T - 1.108 \times 10^{-5} T^2 - 28.22 \times 10^{-8} T^3$$
 (16)

$$\hat{c}_{p,N_2} = 31.5 - 1.357 \times 10^{-2} T + 2.68 \times 10^{-5} T^2 - 1.168 \times 10^{-8} T^3$$
 (17)

$$\hat{c}_{p,O_2} = 28.11 - 0.000368 \times 10^{-2} T + 1.746 \times 10^{-5} T^2 - 1.065 \times 10^{-8} T^3 (18)$$

$$\hat{c}_{p,Ar} = 20.8$$
 (19)

$$\hat{c}_{p,CO_2} = 19.8 + 7.344 \times 10^{-2} T - 5.602 \times 10^{-5} T^2 + 1.715 \times 10^{-8} T^3$$
 (20)

The sensible heat of enthalpy (kJ/s) can be defined as [10]

$$\sum_{i=1}^{m} \hat{n}_{i} \int_{T_{0}}^{T} \hat{c}_{pi} dT$$
(21)

where, i = number of component,  $\hat{n}_i$  is the molar flow rate of the i<sup>th</sup> component (kmol/s) and  $\hat{c}_{p,i}$  is the molar physical enthalpy (kJ/kmol) of the i<sup>th</sup> component.

#### Exit Temperatures of Air Compressor and Gas Turbine

The compressor exit temperature can be calculated by [10]

$$T_{g^2} = T_{g^1} + \frac{T_{g^1}}{\eta_c} \left[ \left( \frac{P_2}{P_1} \right)^{\frac{\gamma_a - 1}{\gamma_a}} - 1 \right]$$
(22)

The gas turbine exit temperature can be calculated by [10]

$$T_{g4} = T_{g3} - \eta_t T_{g3} \left[ 1 - \left( \frac{p_{g4}}{p_{g3}} \right)^{\frac{\gamma_g - 1}{\gamma_g}} \right]$$
(23)

where  $T_{g1}$ ,  $T_{g3}$ ,  $\eta_c$  and  $\eta_t$  are the compressor inlet air temperature, turbine inlet temperature, compressor efficiency, and turbine efficiency, respectively. The values of specific heat ratio for air and combustion gases are approximated to be equal and taken as 1.4.

#### **Performance Parameters**

The net power generation (kW) of the combined cycle can be expressed as

$$\widehat{W}_{cc} = (\widehat{W}_{t,gas} - \widehat{W}_{c,sir}) + (\widehat{W}_{t,st} - (\widehat{W}_{CEP} - \widehat{W}_{HPFP} - \widehat{W}_{LPFP})_{p,water})$$
(25)

The specific fuel consumption (SFC) can be expressed as

$$\mathscr{O}_f = \frac{3600m_f}{\dot{W}_{net,GT}} \tag{26}$$

The overall energy efficiency of the combined cycle plant can be expressed as

$$\eta_{I,CC} = \frac{\widehat{W}_{\alpha}}{\widehat{E}_{d}}$$
(27)

The overall exergy efficiency of the combined cycle plant can be expressed as

$$\eta_{II,CC} = \frac{\widehat{W}_{\alpha}}{\widehat{X}_{f}}$$
(28)

#### **OPTIMIZATION METHODOLOGY**

The methodology involves four steps:

- Develop a design of experiment (DOE) by selecting different levels and the input factor. For the present study, the number of levels opted are three and the input parameters are seven, hence, L<sub>27</sub> (3^7) orthogonal array has been chosen for the design summary using the Taguchi method.
- In the second step, input experimental data and desired output are collected.
- Apply the optimization technique.
- In the end, output results are validated using the artificial neural network (ANN) approach

#### **Taguchi Analysis**

The Taguchi technique, first explained by Dr. Genichi Taguchi, is an effective statistical method extensively adopted in engineering systems to perform optimization using an orthogonal array table with a minimum count of experiments. This approach investigates the effect of considered design parameters on the different response variables using Signal-to-Noise (S/N) analysis.

### **Grey Relational analysis**

This approach evaluates and categorizes the most effective input parameter that gives the best response to the variation. Table 2 describes the list of input parameters with different levels and the response variables. For this, the theoretical response data should be normalized with Larger is better, smaller is better, and nominal is better. The present study maximizes the net power generation, thermal efficiency and exergetic efficiency and minimizes the specific fuel consumption. The 'Larger is better' and the 'Smaller is better' are described in Eq (31) and (32) respectively.

Larger is better

$$y_{i}(f) = \frac{x_{i}(f) - \min x_{i}^{n}(f)}{\max x_{i}^{n}(f) - \min x_{i}^{n}(f)}$$
(31)

Smaller is better

$$y_{i}(f) = \frac{\max x_{i}^{n}(f) - x_{i}(f)}{\max x_{i}^{n}(f) - \min x_{i}^{n}(f)}$$
(32)

Furthermore, the grey relational coefficient (GRC) establishes the correlation between actual and comparable series, which is formulated as,

$$\Psi_i = \frac{\Delta_{\min} + \zeta \Delta_{\max}}{\Delta_i + \zeta \Delta_{\max}}$$
(33)

The grey relational grade (GRG) is then computed as the average of GRC values of all the parameters and can be expressed as

$$GRG = \frac{1}{n} \left( \sum_{y=1}^{n} \alpha_{y} \Psi_{i} \right)$$
(34)

Here  $\sum_{y=1}^{n} \alpha_{y} = 1$ 

## Analysis of Variance (ANOVA)

The significant contribution of each input parameter on the response variable can be determined by calculating the contribution ratio using the ANOVA approach. It has been proved in the literature that the higher the F-value of the input parameter, the more is the impact on the output variable.

Contribution ratio (%) = 
$$\frac{(SS)_f}{(SS)_f}$$
 (35)

where  $SS_f$  refers to sum of square of each input parameter and  $SS_t$  refers to the total sum of square of all the input parameters.

| Input Parameters                         | Code No. of Levels |        |        | Response Variables |                                  |
|--|--------------------|--------|--------|--------------------|----------------------------------|
|  |                    | 1      | 2      | 3                  |                                  |
| Compressor inlet air temperature (K)     | А                  | 307.15 | 305.15 | 303.15             | Specific fuel consumption (SFC); |
| Pressure ratio (r <sub>p</sub> )         | В                  | 10.2   | 11.5   | 13                 | Net Power generation;            |
| Fuel temperature (K)                     | С                  | 309.15 | 312.15 | 315.15             | Thermal efficiency;              |
| Fuel temperature (m <sup>3</sup> /hr)    | D                  | 40444  | 40861  | 40664              | Exergetic Efficiency;            |
| Gas Turbine maximum temperature (K)      | Е                  | 1327.2 | 1450   | 1550               |                                  |
| Compressor efficiency $(\eta_{com})$ (%) | F                  | 88     | 90     | 92                 |                                  |
| Turbine efficiency $(\eta_{tur})$ (%)    | G                  | 77     | 79     | 82                 |                                  |

Table 2. Description of input parameters with different levels

The working procedure of ANN is equivalent to the human brain neuron network. Each neuron retrieves and saves the experimental observations during its training session. Then the neuron network is trained and simulated with the optimal group of variables of the grey Taguchi-based analysis. Neurons are associated with one another by their synaptic weights that function as per the nature of activation function such as logsis tansig or purelin to assess the needed outcome for a given input argument [35]. In MATLAB, the 'nntool' command builds, train and simulate an ANN that will help to validate the best possible collection of input factors obtained from the grey relational grade. If the expected and optimum experimental values are close to each other, then the efficacy of the optimum combination can be validated and authenticated. This will provide optimum values of input parameters to the design engineer and the operators of the power plant.

## **RESULTS AND DISCUSSION**

Under this segment, the GRA is implemented to optimize the multi-response variable by calculating its grey relational coefficient and converting them into a single response value by evaluating its grey relational grade, an illustration of which is explained in the above section. Thereafter, the contribution and impact of each parameter are analyzed using the ANOVA method. In the end, the

| Runs | s Orthogonal Array Design |   |   |   |   |   |   | Output Resp          | onse        |                 |              |
|------|---------------------------|---|---|---|---|---|---|----------------------|-------------|-----------------|--------------|
|      | A                         | В | С | D | E | F | G | W <sub>cc</sub> (kW) | $\eta_{th}$ | η <sub>ex</sub> | SFC (kg/kWh) |
|      |                           |   |   |   |   |   |   |                      | (%)         | (%)             |              |
| 1    | 1                         | 1 | 1 | 1 | 1 | 1 | 1 | 212471               | 53.738      | 54.175          | 0.26990      |
| 2    | 1                         | 1 | 1 | 1 | 2 | 2 | 2 | 245501               | 61.719      | 62.597          | 0.20368      |
| 3    | 1                         | 1 | 1 | 1 | 3 | 3 | 3 | 259911               | 64.994      | 66.271          | 0.18398      |
| 4    | 1                         | 2 | 2 | 2 | 1 | 1 | 1 | 221311               | 55.936      | 56.956          | 0.24845      |
| 5    | 1                         | 2 | 2 | 2 | 2 | 2 | 2 | 239001               | 60.029      | 61.509          | 0.21414      |
| 6    | 1                         | 2 | 2 | 2 | 3 | 3 | 3 | 254461               | 63.603      | 65.487          | 0.19108      |
| 7    | 1                         | 3 | 3 | 3 | 1 | 1 | 1 | 210301               | 53.806      | 32.111          | 0.27202      |
| 8    | 1                         | 3 | 3 | 3 | 2 | 2 | 2 | 229651               | 58.525      | 35.066          | 0.22770      |
| 9    | 1                         | 3 | 3 | 3 | 3 | 3 | 3 | 246341               | 62.463      | 37.614          | 0.19964      |
| 10   | 2                         | 1 | 2 | 3 | 1 | 2 | 3 | 249641               | 63.401      | 64.996          | 0.19677      |
| 11   | 2                         | 1 | 2 | 3 | 2 | 3 | 1 | 240891               | 60.802      | 62.718          | 0.21001      |
| 12   | 2                         | 1 | 2 | 3 | 3 | 1 | 2 | 246621               | 61.939      | 64.210          | 0.20114      |
| 13   | 2                         | 2 | 3 | 1 | 1 | 2 | 3 | 240041               | 61.868      | 64.067          | 0.20825      |
| 14   | 2                         | 2 | 3 | 1 | 2 | 3 | 1 | 232591               | 59.576      | 62.078          | 0.22099      |
| 15   | 2                         | 2 | 3 | 1 | 3 | 1 | 2 | 238621               | 60.828      | 63.688          | 0.21056      |
| 16   | 2                         | 3 | 1 | 2 | 1 | 2 | 3 | 234966               | 58.826      | 59.323          | 0.22325      |
| 17   | 2                         | 3 | 1 | 2 | 2 | 3 | 1 | 228781               | 56.911      | 57.761          | 0.23496      |
| 18   | 2                         | 3 | 1 | 2 | 3 | 1 | 2 | 235541               | 58.308      | 59.468          | 0.22222      |
| 19   | 3                         | 1 | 3 | 2 | 1 | 3 | 2 | 241451               | 61.615      | 64.239          | 0.20812      |
| 20   | 3                         | 1 | 3 | 2 | 2 | 1 | 3 | 252841               | 64.106      | 67.269          | 0.19142      |
| 21   | 3                         | 1 | 3 | 2 | 3 | 2 | 1 | 239841               | 60.488      | 63.811          | 0.21072      |
| 22   | 3                         | 2 | 1 | 3 | 1 | 3 | 2 | 235571               | 59.284      | 60.557          | 0.22109      |
| 23   | 3                         | 2 | 1 | 3 | 2 | 1 | 3 | 248291               | 62.054      | 63.826          | 0.20063      |
| 24   | 3                         | 2 | 1 | 3 | 3 | 2 | 1 | 237851               | 59.137      | 61.143          | 0.21712      |
| 25   | 3                         | 3 | 2 | 1 | 1 | 3 | 2 | 225341               | 57.548      | 59.399          | 0.23725      |
| 26   | 3                         | 3 | 2 | 1 | 2 | 1 | 3 | 238871               | 60.615      | 62.965          | 0.21217      |
| 27   | 3                         | 3 | 2 | 1 | 3 | 2 | 1 | 229361               | 57.902      | 60.459          | 0.22920      |

Table 3. Orthogonal Array-based experimental results of output response

performance parameters of the considered system are validated based on a neural network technique.

## Formation of Output Response Using OA

Firstly, an orthogonal array (OA) of L27 ( $3^7$ ) was designed to conduct experimentation in Minitab. There is a total of 27 experimental runs by taking seven input parameters namely, compressor inlet air temperature (A), pressure ratio (B.), inlet fuel temperature (C), the volumetric flow rate of fuel (D), inlet gas temperature (E), compressor isentropic efficiency (F), turbine isentropic efficiency (G) and 3 different levels to organize the output response namely, the net power generation ( $W_{cc}$ ), energy efficiency ( $\eta_{en}$ ), exergy efficiency ( $\eta_{ex}$ ), and specific fuel consumption (SFC) as stated in Table 3.

From the experimentation results, it is noted that the decrease in compressor inlet air temperature decreases the power consumed by the compressor, increases the power developed by the turbine and hence, increases the net power output of the gas turbine cycle. The reason for the increase in power developed by the turbine with lower compressor inlet air temperature being that a decrease in air temperature increases the density of air and consequently, increases the mass flow rate [36]. The increase in net power output with the same fuel flow rate decreases the specific fuel consumption. On the other hand, compressor pressure ratio has an adverse effect on the net power output. When the compressor pressure ratio increases with an increase in turbine inlet temperature, the net power output decreases due to increase in the compressor work.

However, the overall thermal efficiency increases with a higher compression ratio [37,38]. Increasing the fuel temperature reduces the fuel consumption rate required for the same gas turbine inlet temperature and thereby reduces the power output of the cycle. But, the overall thermal efficiency increases with an increase in the temperature of the fuel. The isentropic efficiency describes the efficacy of the turbine and compressor and affects the overall performance of the system [39].

#### Grey Taguchi Analysis

## Normalization of output response

The comparative analysis for different response variables cannot be made as they are assigned with different dimensions. So, it is important to alter them in dimensionless quantity and hence normalization for output response data was performed for the objective function as mentioned in Table 4. The value of normalized data lies between 0 and 1 and values closer to 1 suggest better performance characteristics [40]

# Computation of Grey Relational Coefficient (GRC) and Grey Relational Grade (GRG)

Subsequently, a quality loss function (QLF) or deviation of response variables from actual series is evaluated which is a measure of the variation of actual from comparable series. The smaller the value of QLF, the closer is the ideal approach condition with the minimum loss [41]. The GRC can be determined using the QLF and develops correlation among actual and comparable series. The transformation of the multi-objective function into a single response variable

| Runs | Normaliza       | tion of Outpu | ıt response |                   | Runs | Normalization of Output response |                |             |                   |  |
|------|-----------------|---------------|-------------|-------------------|------|----------------------------------|----------------|-------------|-------------------|--|
|      | Larger is       | better        |             | Smaller is better | _    | Larger is b                      | etter          |             | Smaller is better |  |
|      | W <sub>cc</sub> | $\eta_{th}$   | $\eta_{ex}$ | SFC               | _    | W <sub>cc</sub>                  | $\eta_{_{th}}$ | $\eta_{ex}$ | SFC               |  |
| 1    | 0.04374         | 0.00000       | 0.62756     | 0.02400           | 15   | 0.57085                          | 0.62985        | 0.89813     | 0.69808           |  |
| 2    | 0.70953         | 0.70906       | 0.86711     | 0.77628           | 16   | 0.49718                          | 0.45201        | 0.77397     | 0.55396           |  |
| 3    | 1.00000         | 1.00000       | 0.97162     | 1.00000           | 17   | 0.37251                          | 0.28184        | 0.72956     | 0.42092           |  |
| 4    | 0.22193         | 0.19525       | 0.70666     | 0.26775           | 18   | 0.50877                          | 0.40598        | 0.77810     | 0.56566           |  |
| 5    | 0.57851         | 0.55892       | 0.83615     | 0.65744           | 19   | 0.62790                          | 0.69980        | 0.91381     | 0.72580           |  |
| 6    | 0.89014         | 0.87638       | 0.94932     | 0.91937           | 20   | 0.85749                          | 0.92112        | 1.00000     | 0.91549           |  |
| 7    | 0.00000         | 0.00601       | 0.00000     | 0.00000           | 21   | 0.59544                          | 0.59967        | 0.90162     | 0.69629           |  |
| 8    | 0.39004         | 0.42524       | 0.08404     | 0.50344           | 22   | 0.50937                          | 0.49270        | 0.80907     | 0.57843           |  |
| 9    | 0.72647         | 0.77512       | 0.15652     | 0.82214           | 23   | 0.76577                          | 0.73881        | 0.90208     | 0.81093           |  |
| 10   | 0.79299         | 0.85846       | 0.93535     | 0.85479           | 24   | 0.55533                          | 0.47968        | 0.82574     | 0.62353           |  |
| 11   | 0.61661         | 0.62755       | 0.87055     | 0.70436           | 25   | 0.30317                          | 0.33847        | 0.77614     | 0.39491           |  |
| 12   | 0.73211         | 0.72855       | 0.91298     | 0.80506           | 26   | 0.57589                          | 0.61094        | 0.87758     | 0.67982           |  |
| 13   | 0.59948         | 0.72226       | 0.90891     | 0.72438           | 27   | 0.38420                          | 0.36990        | 0.80628     | 0.48635           |  |
| 14   | 0.44931         | 0.51865       | 0.85235     | 0.57958           |      |                                  |                |             |                   |  |

#### Table 4. Normalization of output response

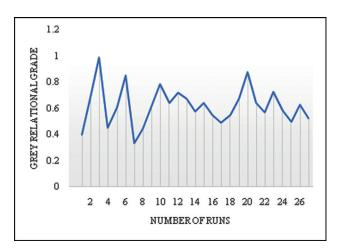
| Runs | Quality Lo      | oss function ( | (Δ)         |         | Gray Relat      | tional Coeffic | cient ( $\zeta = 0.5$ ) | )       | Grade (GRG) | Rank |
|------|-----------------|----------------|-------------|---------|-----------------|----------------|-------------------------|---------|-------------|------|
|      | W <sub>cc</sub> | $\eta_{th}$    | $\eta_{th}$ | SFC     | W <sub>cc</sub> | $\eta_{th}$    | $\eta_{th}$             | SFC     |             |      |
| 1    | 0.95626         | 1.00000        | 0.37243     | 0.97600 | 0.34335         | 0.33333        | 0.57311                 | 0.33875 | 0.39714     | 26   |
| 2    | 0.29047         | 0.29094        | 0.13289     | 0.22372 | 0.63254         | 0.63216        | 0.79003                 | 0.69088 | 0.68640     | 7    |
| 3    | 0.00000         | 0.00000        | 0.02838     | 0.00000 | 1.00000         | 1.00000        | 0.94629                 | 1.00000 | 0.98657     | 1    |
| 4    | 0.77807         | 0.80475        | 0.29334     | 0.73225 | 0.39122         | 0.38322        | 0.63025                 | 0.40576 | 0.45261     | 24   |
| 5    | 0.42149         | 0.44108        | 0.16385     | 0.34256 | 0.54260         | 0.53130        | 0.75319                 | 0.59343 | 0.60513     | 15   |
| 6    | 0.10986         | 0.12363        | 0.05068     | 0.08064 | 0.81986         | 0.80176        | 0.90797                 | 0.86113 | 0.00000     | 3    |
| 7    | 1.00000         | 0.99399        | 1.00000     | 1.00000 | 0.33333         | 0.33467        | 0.33333                 | 0.33333 | 0.33367     | 27   |
| 8    | 0.60996         | 0.57476        | 0.91596     | 0.49656 | 0.45047         | 0.46522        | 0.35312                 | 0.50173 | 0.44263     | 25   |
| 9    | 0.27353         | 0.22488        | 0.84348     | 0.17787 | 0.64638         | 0.68977        | 0.37217                 | 0.73761 | 0.61148     | 14   |
| 10   | 0.20702         | 0.14154        | 0.06465     | 0.14521 | 0.70720         | 0.77938        | 0.88550                 | 0.77495 | 0.78675     | 4    |
| 11   | 0.38339         | 0.37245        | 0.12945     | 0.29564 | 0.56600         | 0.57310        | 0.79434                 | 0.62842 | 0.64047     | 12   |
| 12   | 0.26789         | 0.27146        | 0.08702     | 0.19494 | 0.65114         | 0.64813        | 0.85176                 | 0.71949 | 0.71763     | 6    |
| 13   | 0.40052         | 0.27774        | 0.09109     | 0.27562 | 0.55523         | 0.64289        | 0.84589                 | 0.64465 | 0.67216     | 9    |
| 14   | 0.55070         | 0.48135        | 0.14765     | 0.42042 | 0.47588         | 0.50950        | 0.77203                 | 0.54323 | 0.57516     | 17   |
| 15   | 0.42915         | 0.37015        | 0.10187     | 0.30192 | 0.53813         | 0.57461        | 0.83074                 | 0.62351 | 0.64175     | 10   |
| 16   | 0.50282         | 0.54799        | 0.22603     | 0.44604 | 0.49859         | 0.47710        | 0.68868                 | 0.52852 | 0.54822     | 19   |
| 17   | 0.62749         | 0.71816        | 0.27044     | 0.57908 | 0.44346         | 0.41046        | 0.64898                 | 0.46336 | 0.49156     | 23   |
| 18   | 0.49123         | 0.59402        | 0.22190     | 0.43434 | 0.50442         | 0.45703        | 0.69262                 | 0.53514 | 0.54730     | 20   |
| 19   | 0.37210         | 0.30020        | 0.08619     | 0.27420 | 0.57333         | 0.62484        | 0.85296                 | 0.64583 | 0.67424     | 8    |
| 20   | 0.14251         | 0.07888        | 0.00000     | 0.08451 | 0.77820         | 0.86373        | 1.00000                 | 0.85542 | 0.87434     | 2    |
| 21   | 0.40456         | 0.40033        | 0.09838     | 0.30371 | 0.55276         | 0.55535        | 0.83560                 | 0.62211 | 0.64145     | 11   |
| 22   | 0.49063         | 0.50730        | 0.19093     | 0.42157 | 0.50473         | 0.49638        | 0.72367                 | 0.54255 | 0.56683     | 18   |
| 23   | 0.23423         | 0.26119        | 0.09792     | 0.18907 | 0.68099         | 0.65686        | 0.83623                 | 0.72562 | 0.72493     | 5    |
| 24   | 0.44467         | 0.52032        | 0.17426     | 0.37647 | 0.52929         | 0.49004        | 0.74156                 | 0.57047 | 0.58284     | 16   |
| 25   | 0.69684         | 0.66153        | 0.22386     | 0.60509 | 0.41777         | 0.43047        | 0.69074                 | 0.45245 | 0.49786     | 22   |
| 26   | 0.42411         | 0.38907        | 0.12242     | 0.32019 | 0.54106         | 0.56239        | 0.80332                 | 0.60962 | 0.62910     | 13   |
| 27   | 0.61580         | 0.63010        | 0.19372     | 0.51365 | 0.44811         | 0.44244        | 0.72076                 | 0.49327 | 0.52614     | 21   |

Table 5. Estimation of parameters of Grey relational-Taguchi method

is done with the help of gray relational grade (GRG) [42]. With reference to OA and GRC values, the GRG and rank for different experiments can be evaluated using Eq. (34) and demonstrated in Table 5. Higher count of GRG implies that input conditions and output responses are close to an optimal solution.

## Optimal combinations of parameters

The most favorable association of input parameters is analyzed from GRG values for each experiment. The higher value of grade ensures closer to optimality. The graphical representation of GRG for different experiment runs is presented in Fig. 2. and indicates the impact and influence of each parameter on output results. Fig. 2. reveals that Case 3 and Case 7 have the highest and lowest value of GRG among 27 runs. Taking into account all combinations, the optimum input conditions for the most favorable performance of a

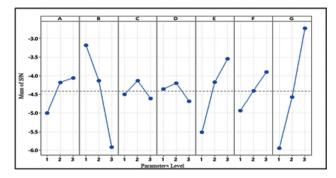


**Figure 2.** Variation of grey relational grade for different number of runs.

combined gas-steam plant are determined as compressor inlet air temperature = 307.15 K, pressure ratio = 10.2, fuel temperature = 309.15 K, volumetric flow rate of fuel =40444 m<sup>3</sup>/hr, gas turbine maximum temperature =1550 K, compressor efficiency = 92 %, turbine efficiency = 82 %. Under this condition ( $A_1B_1C_1D_1E_3F_3G_3$ ), the optimum power generated, thermal efficiency, exergetic efficiency, and specific fuel consumption comes out to be 259911 kW, 64.9 %, 66.27 %, and 0.1839 kg/kWh respectively. Furthermore, the condition  $A_1B_3C_3D_3E_1F_1G_1$  minimizes the system performance

Table 6. Grey relational grade-based response table

| Level | Α      | В      | С      | D      | Ε      | F      | G      |
|-------|--------|--------|--------|--------|--------|--------|--------|
| 1     | -4.997 | -3.188 | -4.494 | -4.353 | -5.512 | -4.933 | -5.934 |
| 2     | -4.174 | -4.134 | -4.131 | -4.195 | -4.171 | -4.399 | -4.566 |
| 3     | -4.058 | -5.908 | -4.604 | -4.682 | -3.547 | -3.898 | -2.730 |
| Delta | 0.939  | 2.720  | 0.473  | 0.487  | 1.965  | 1.035  | 3.205  |
| Rank  | 5      | 2      | 7      | 6      | 3      | 4      | 1      |



**Figure 3.** Multiple response characteristic for different input parameters level.

with conditions as follow, compressor inlet air temperature = 307.15 K, pressure ratio = 13, fuel temperature = 315.15 K, volumetric flow rate of fuel = 40664 m<sup>3</sup>/hr, gas turbine maximum temperature = 1327.15 K, compressor efficiency = 88 %, turbine efficiency = 77 % respectively. The net power generated, thermal efficiency, exergetic efficiency, and specific fuel consumption at this state are 259911 kW, 64.9 %, 66.27 %, 0.1839 kg/kWh and 210301 kW, 53.8 %, 32.1 %, 0.2720 kg/kWh, respectively.

Response table is developed for different factors using GRG to pinpoint the impact of all parameters on the response variable as displayed in Table 6. The arrangement of precedence of input parameters for the optimum performance of the system is G>B>E>F>A>D>C. The maximum and minimum response of parameters for different levels are shown in Fig. 3.

## Analysis of Variance (ANOVA)

Applying the ANOVA method, the significant contribution of each parameter is examined and analyzed based on multiple objective functions as displayed in Table 7 [43,44]. It can be seen from the findings that the impact of turbine efficiency on the performance characteristic is maximum with a contribution ratio of 42.41 %, followed by pressure ratio (31.69 %), gas turbine maximum temperature (13.99 %), compressor efficiency (3.43 %), compressor inlet air temperature (1.33 %), the volumetric flow rate of fuel (0.81 %) and fuel temperature (0.59 %) which are in accord with the response table using GRG. The literature [45,46] reveals that if the probability (P-value) of any parameter is less than 0.05, the parameter is significant. In the results, the P-value for turbine efficiency and pressure ratio is 0.000, hence contributes maximum.

It is clearly understandable from the ANOVA findings that turbine efficiency is the most important parameter to be considered for the performance of the CC. The effect of turbine efficiency on system performance has also been

Table 7. Contribution of each parameter on output variables using grey relational grade

| Parameters                       | DF | Adj SS  | Adj MS  | F-Value | P-Value | Contribution Ratio (%) |
|----------------------------------|----|---------|---------|---------|---------|------------------------|
| Compressor inlet air temperature | 2  | 0.00746 | 0.00373 | 1.400   | 0.284   | 1.33415                |
| Pressure ratio                   | 2  | 0.17716 | 0.08858 | 33.250  | 0.000   | 31.69150               |
| Fuel Temperature                 | 2  | 0.00332 | 0.00166 | 0.620   | 0.553   | 0.59355                |
| Volumetric flow rate of fuel     | 2  | 0.00455 | 0.00227 | 0.850   | 0.450   | 0.81341                |
| Gas turbine inlet temperature    | 2  | 0.07823 | 0.03912 | 14.690  | 0.001   | 13.99515               |
| Compressor efficiency            | 2  | 0.01922 | 0.00961 | 3.610   | 0.059   | 3.43805                |
| Turbine efficiency               | 2  | 0.23711 | 0.11856 | 44.510  | 0.000   | 42.41621               |
| Error                            | 12 | 0.03196 | 0.00266 |         |         |                        |
| Total                            | 26 | 0.55901 |         |         |         |                        |

| -                  | enberg-Mar<br>an Squared | derand)<br>rquardt (trainlm)<br>Error (mse) |          |
|--------------------|--------------------------|---|----------|
| Progress           |                          |   |          |
| Epoch:             | 0                        | 9 iterations                                | 1000     |
| Time:              | Ì                        | 0:00:01                                     |          |
| Performance:       | 0.0715                   | 2.79e-07                                    | 0.00     |
| Gradient:          | 0.129                    | 7.44e-05                                    | 1.00e-07 |
| Mu:                | 0.00100                  | 1.00e-06                                    | 1.00e+10 |
| Validation Checks: | 0                        | 6   | 6        |
| Plots              |                          |   |          |
| Performance        | ) (plotperf              | form)                                       |          |
| Training State     | (plottrain               | nstate)                                     |          |
| Regression         | ) (plotregr              | ression)                                    |          |
| Plot Interval:     |                          | 1 epoch                                     | is       |

| Training Info Training Par<br>Training Data |         |   | Training Results         |                            |
|---|---------|---|--------------------------|----------------------------|
| Inputs                                      | input   | v | Outputs                  | ANNgreytaguchi_outputs     |
| Targets                                     | target  | v | Errors                   | ANNgreytaguchi_errors      |
| Init Input Delay States                     | (zeros) | ~ | Final Input Delay States | ANNgreytaguchi_inputStates |
| Init Layer Delay States                     | (zeros) | V | Final Layer Delay States | ANNgreytaguchi_layerStates |
|   |         |   |                          |                            |
|   |         |   |                          |                            |
|   |         |   |                          |                            |

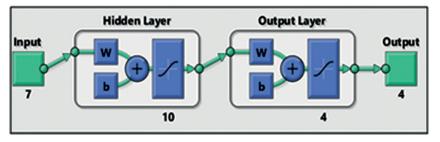


Figure 4. Modelling and Training using Artificial Neural Network Technique.

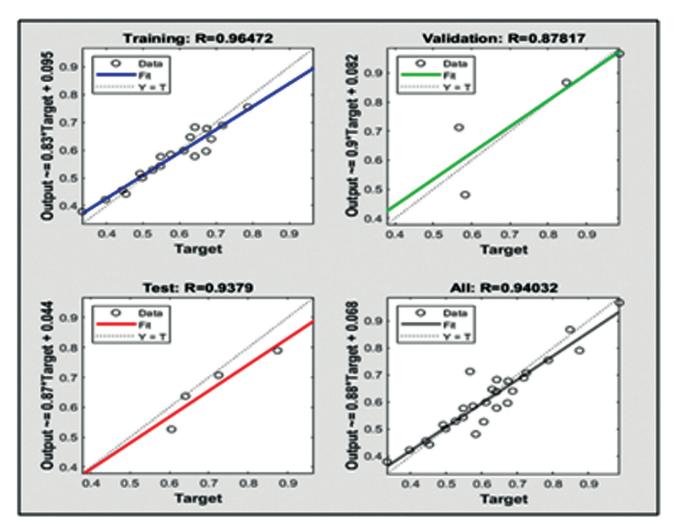


Figure 5. Regression plot for Neural Network.

studied by Siddiqui and Dincer [47], and Ibrahim and Rahman [48] and reveals that higher lefficiency of the turbine increases the power and thermodynamic efficiency of the overall system. It is also obvious from the results that fuel parameters are less important while considering the multiple responses of the system.

## Validation of Grey-Taguchi analysis results using Artificial neural network (ANN)

Finally, the neural network simulating methodology is opted to verify and validate the optimal response obtained by the Grey-Taguchi method. Numerous researchers [31,49–51] use this technique for the validation of output results of engineering applications. With the aid of the 'nntool' command and importing input and target parameters from Table 2, the ANN model is created, trained, and simulated using the feed-forward backprop technique with the number of neurons layers to be 3 namely, input layers (7 neurons), hidden layer (10 neurons) and output layer (4 neurons) as shown in Fig. 4. The selection of neurons for the hidden layer is undertaken using the trial-and-error method. In the hidden layer, the number of neurons can be changed in the course of training until TANSIG (tangent sigmoid) reduces the mean square error to 0.0527. The coefficient of Regression (R) measures the degree of deviation of predicted and actual values as depicted in Fig. 5 and reveals that the data of predicted values are closely related to the grey Taguchi results. Furthermore, it turns out from Fig. 6 that the optimal output response for the ANN criterion is obtained at epoch 3. Also, Fig 7. compares the output result obtained from grey Taguchi and ANN method with GRG as target parameter and indicates that the results obtained from both the methods show optimality at Run 3. This validates the potential and effectiveness of the grey Taguchi analysis and neural network procedure.

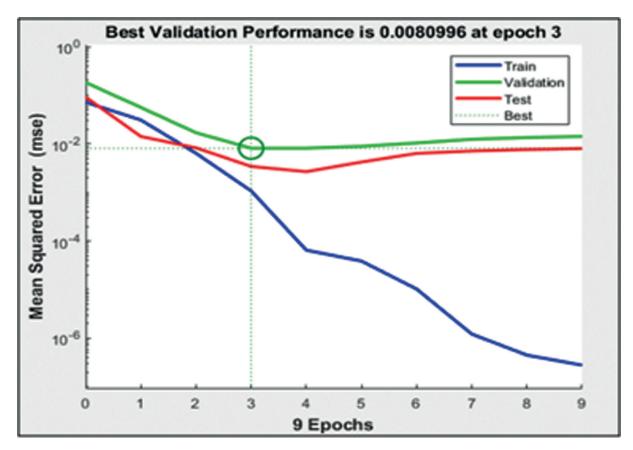


Figure 6. Performance plot for optimal conditions.

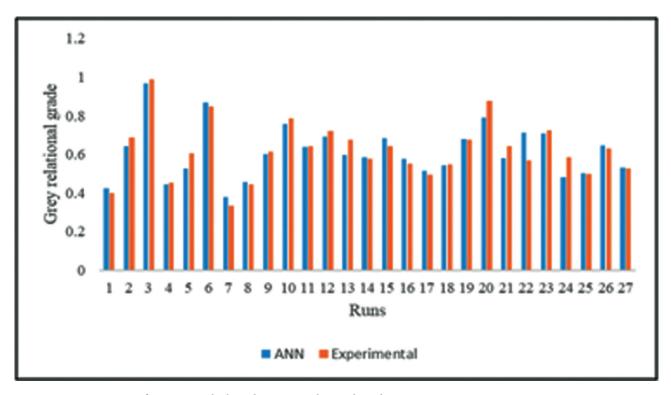


Figure 7. Comparison of output results based on Grey relational grade.

## CONCLUSIONS

This research work is aimed to investigate the optimal combination of input parameters and their contribution to achieve the multi-objective optimization of a combined gas-steam-based power plant using Taguchi-based grey relational analysis. In this investigation, compressor inlet air temperature, pressure ratio, fuel temperature, volumetric flow rate of fuel, gas turbine maximum temperature, compressor efficiency, and turbine efficiency have been considered. The impact of aforesaid parameters on net power generation, thermal efficiency, exergetic efficiency, and specific fuel consumption was examined. The major findings of the work are as follows.

- The optimum input conditions for the most favorable performance of the combined gas-steam plant under study are obtained as: compressor inlet air temperature = 307.15 K, pressure ratio = 10.2, fuel inlet temperature = 309.15 K, volumetric flow rate of fuel =40444 m<sup>3</sup>/hr, gas turbine maximum temperature =1550 K, compressor efficiency = 92 %, turbine efficiency = 82 %.
- Under the maximum (A<sub>1</sub>B<sub>1</sub>C<sub>1</sub>D<sub>1</sub>E<sub>3</sub>F<sub>3</sub>G<sub>3</sub>) and minimum (A<sub>1</sub>B<sub>3</sub>C<sub>3</sub>D<sub>3</sub>E<sub>1</sub>F<sub>1</sub>G<sub>1</sub>) conditions, the net power generated, thermal efficiency, exergetic efficiency, and specific fuel consumption come out to be 259911 kW, 64.9%, 66.27%, 0.1839 kg/kWh and 210301 kW, 53.8%, 32.1%, 0.2720 kg/kWh respectively.
- ANOVA analysis reveals that the impact of turbine efficiency on the performance characteristic is maximum with a contribution ratio of 42.41 %, followed by pressure ratio (31.69 %), gas turbine (13.99 %), compressor efficiency (3.43 %), ambient temperature (1.33 %), the volumetric flow rate of fuel (0.81 %) and the fuel temperature (0.59 %). These results are in accordance with the response table obtained using grey relation grade.
- The optimal output response values obtained from grey Taguchi analysis are close to the predicted values of artificial neural network and reveals that both methods have high potential and efficacy to yield optimal results.

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## **AUTHORSHIP CONTRIBUTIONS**

Authors equally contributed to this work.

## DATA AVAILABILITY STATEMENT

The authors confirm that the data that supports the findings of this study are available within the article. Raw data that support the finding of this study are available from the corresponding author, upon reasonable request.

## CONFLICT OF INTEREST

The author declared no potential conflicts of interest with respect to the research, authorship, and/or publication of this article.

## **ETHICS**

There are no ethical issues with the publication of this manuscript.

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